Oil-Free Centrifugal Hydrogen Compression Technology Demonstration

PI: Hooshang Heshmat, PhD Mohawk Innovative Technology, Inc. May 10, 2011

Project ID # PD016

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Overview

Timeline

- Start Sept 1, 2008
- Funding Authorized 2/28/09
- End Aug 31, 2011
- Ext. Requested to Aug 2012
- 48 % Complete

Budget

- Total Project Funding
 - \$2,992,407 DOE
 - \$1,149,253 MiTi/MHI Cost Share
- \$346,208 FY10 Funding
- \$720,000 FY11 Funding

Barriers

- Hydrogen Delivery Compressor
 - Reliability
 - System Cost
 - Efficiency of H₂ Gas Compression

Partners

- Lead: Mohawk Innovative Technology, Inc. Mohawk Innovative Technology, Inc.
 - Albany, NY
- Mitsubishi Heavy Industry



Hiroshima, Japan

Project Objectives

- Relevance

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Design a reliable and cost effective centrifugal compressor for hydrogen pipeline transport

- Flow 240,000 to 500,000 kg/day
- Pressure Rise to 300-500 psig up to 1,200-1,500 psig
- Contaminant-Free/Oil-Free Hydrogen

| | | Project Target | |
|--|------------------|-------------------|--------|
| Category | 2005 Status | FY2012 | FY2017 |
| Reliability | Low | Improved | High |
| Energy Efficiency | 98% | 98% | >98% |
| Capital Investment (\$M) (based on 200,000 kg of H2/day) | \$15 | \$12 | \$9 |
| Maintenance (% of Total Capital Investment) | 10% | 7% | 3% |
| Contamination | Varies by Design | | None |

Hydrogen, Fuel Cells & Infrastructure Technologies Program [DOE Publication]

Compressor Design Methodology

- Approach

Compressor Design Analysis

• Mean Line Analysis, CFD, FEA

Sub-Component Design

- Foil Bearings & Seals
- Coatings

Design Single-Stage

- Impeller, Diffuser and Others
- Drive System & Test Loop

Single Stage Proof Testing

Fabricate & Characterize Pressure & Flow

Scale System Design

- Predict Complete System Performance
- Update Multi-Stage, Multi-Frame Design
- Economic Analysis









Demonstrate feasibility of very high speed hydrogen centrifugal compressor

MiTi[®] Foil Bearings – Operational Concept

- Approach

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Compliant Foil Seals

6 - Approach

- Derived From MiTi's Foil Bearing Technology
- Both Face and Shaft Sealing Available
- Non-Contacting
- Capable of Accommodating Large Excursions and Extreme Environments
- Very Low Leakage
 Compared to State-of-the-Art Sealing Alternatives*



* Salehi, M. and Heshmat, H. "High Temperature Performance Evaluation of a Compliant Foil Seal," Paper No. AIAA 2000-3376, Proceedings of the 36th Joint Propulsion Conference and Exhibit, July 16-19, 2000, Huntsville, Alabama.

MiTi Korolon[®] Coatings

Approach

Successful Operation of Foil Bearings/Seals Require Solid Lubricant Coatings Korolon[®] Coatings Have Been Specifically Designed for Foil Bearings/Seals



Flexible Ceramic Coating



New Foil Pad



Typical Friction/Speed Results Showing Hydrodynamic Lift

| | Korolon [®] 700 | Korolon [®] 800 & 900 | Korolon [®] 1350 A & B |
|-------------------------|--|--|---|
| Chemical Composition | Polymer based with solid lubricants | Tungsten Disulfide based with solid lubricants | Nickel- Chrome with solid lubricants |
| Service Temperature | Up to 700 °F | Up to 900 °F | 200-1350°F |

- Low Friction and Wear Rate

- Deposited with Spray Gun Process at Room Temp

Recent Research Has Shown That Polyamide Coatings, Such as Korolon[®], are Compatible With H₂ [Int'l Hydrogen Energy Development Forum, Fukuoka, Japan 2011]

MiTi Compressor Design Analysis

- Technical Accomplishments & Progress

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Double-Entry Multi-Frame Compressor

- 6 and 9 Stages (2 and 3 Frames)

Exit Pressure > 1,200 psi

□ Power: 7,800 – 12,000 HP

□ Tip Speed: 1,500 – 2,000 ft/s

Mass Flow: 240K – 500K kg/day

Max Bearing/Seal Temp: 180-200°F

□ Specific Energy: 0.48-0.59 (^{kW-HR}/_{Kg})



Fully Assembled, 3-Frame System

Mitsubishi Compressor Design Analysis

- 9 Technical Accomplishments & Progress
 - Single-Entry Multi-Frame Compressor
 - 7 and 9 Stages (2 and 3 Frames)
 - Design Developed by MHI with Input from MiTi
 - Bearing/Rotor Design Analysis Performed by MiTi
 - Design Based on NG Pipeline Compressors
 - Exit Pressure: > 1,200 psi
 - Power: 8,300 12,000 HP
 - Tip Speed: > 2,000 ft/s
 - Mass Flow: 240K 500K kg/day
 - Specific Energy: 0.44-0.65 $\left(\frac{kW-HR}{Kq}\right)$



Multi-Stage Compressor Design

10 - Technical Accomplishments & Progress

Design Comparison: No. Stages vs. Tip Speed for Three Different Pressure Ratios

- MiTi Double-Entry
- MHI Single-Entry
- Excellent Correlation
 Between the Two
 Designs Within the
 Operating Range



Rotor/Bearing System Dynamics



Rotordynamic Analysis

12 - Technical Accomplishments & Progress

- Centrifugal Compression of H₂ is Only Feasible at Very High Speed
- The Required Operating Speed is Above the Rotor's 1st Bending Frequency
- Foil Bearings/Seal Technology (Stiffness, Damping & Load Capacity) Allows for Safe Operation Above the 1st Bend
- The Operating Speed Range is Free of Resonant Vibration



Multi-Frame Centrifugal Compressor Designs

13 - Technical Accomplishments & Progress

| Design Strategy | Mohawk Innovative | MHI 📩 | |
|---|-------------------|----------------|--|
| Compressor Type | Double-Entry | Single-Entry | |
| Number of Stages | 6 and 9 | 7 and 9 | |
| Number of Frames | 2 and 3 | 2 and 3 | |
| Flow Capacity (Kg H ₂ /day) | 240,000 – 500,000 | | |
| Total Pressure Ratio | 2.4 - 3.33 | | |
| Total Power Input (HP) | 7,800-12,000 | 8,300 – 12,000 | |
| Max Tip Speed (1000 ft/s) | 1.6 – 1.8 | 1.8 - 2.0 | |
| Compressor Footprint (ft ²) | 145 - 160 | 150 - 175 | |

Compressor Design Meets DOE Target

14 - Technical Accomplishments & Progress

| Characteristics | DOE Target | MiTi Estimates |
|---|---------------------------------------|---|
| Efficiency (%) | 98% | 98% |
| Hydrogen Capacity Target (kg/day) | 200,000 | 240,000 – 500,000 |
| Hydrogen Leakage | <0.5 | 0.2 |
| Hydrogen Purity | 99.99 | 99.99 |
| Inlet Pressure (psig) | 300-700 | 350-500 |
| Discharge Pressure (psig) | 1,000-1,200 | 1,226 - 1,285 |
| Total Compressor Package (\$Million) | \$15.6 | \$7.3-\$12.5* |
| Maintenance Cost (% total Capital Investment) | 3% | <3% |
| Annual Maintenance Cost (\$/kW-hr) | \$0.007 | <\$0.005 |
| Package Size (sq-ft) | 300-350 | 145 - 160 |
| Reliability (# of Systems Required) | High - Eliminate Redundant Systems | Very High – Oil-Free Foil Bearings Eliminates Need for Redundant Systems |

* Cost estimates based on data from (1) MHI for comparably sized NG compressor systems,
 (2) published oil and gas industry data and (3) from quotes for fabrication of major components of MiTi's compressor design.

Design of Single Stage Compressor

15 - Technical Accomplishments & Progress

Hermetically **Sealed Chamber** ETT DE at we want Compressor **Coupled Motors** Stage



Single Stage Compressor Test Setup with Two Coupled, Oil-Free Motors

Closed-Loop Test Facility Design

16 - Technical Accomplishments & Progress



Compressor Test Facility Being Fabricated and Assembled at MiTi

Collaborations

17 - Partners and Subcontractors

- Alitsubishi Heavy Industries
 - Single-Entry Centrifugal Compressor Design
 - Computational Fluid Dynamics (CFD)
 - Finite Element Analysis (FEA)
 - » World Leader in NG Pipeline Centrifugal Compressor Systems
 - > Extensive Compressor Test Facilities
 - In-House Turbocompressor Manufacturing



- Material Measurement Laboratory
- Compatibility of Foil Bearing and Foil Seal Materials in H₂ Environment



- International Institute for Carbon-Neutral, Energy Research
 - Consultation on Materials Selection
- Recognized Compressor Design Consultant
- CFD Consulting Group With Compressor Expertise

Proposed Future Work

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Single Stage Compressor Testing

- All Testing to be Conducted in Accordance with Industry Standard ASME PTC-10
- Testing to be Conducted at MiTi
 - Fabricate and Test MiTi® Single-Stage Compressor
 - Validate Compressor with Foil Bearings and Seals
 - Demonstrate Oil-Free Coupling Technology with Foil Bearings
- Testing to be Conducted at Mitsubishi
 - Fabricate and Validate MHI Single-Stage Compressor
- Critical Comparison of the Two Designs
- Hydrogen Compatibility Evaluation of Foil Bearing and Foil Seal Materials at NIST

Summary

- Relevance: The current compression technology used for hydrogen is unreliable, resulting in the need for redundant compressors and thus higher cost. A centrifugal compressor was selected as the most reliable and efficient technology to meet the DOE 2012/2017 performance targets.
- Approach: Design a high-speed, multi-stage, oil-free compressor with MiTi[®] patented foil bearing/seal technology and validate performance by testing a single stage compressor.
- Technical Accomplishments & Progress: Design of a multi-stage single and double entry centrifugal compressor has been completed. Detailed design of a single stage is complete. Foil bearings and seals were fabricated and bench tested. The single stage compressor facility is currently under construction.
- Collaboration: MiTi and Mitsubishi are collaborating on the compressor design and development. NIST, Sandia and the University of Illinois are providing guidance on hydrogen compatibility of materials.
- Proposed Future Work: The single stage testing will be conducted to validate the compressor design. Collaboration with Mitsubishi and others continue as we progress towards the selection of a final hydrogen compressor design.

Acknowledgements

MiTi is grateful for the support from the DOE Hydrogen and Fuel Cells Program and in particular, Sara Dillich, Scott Weil and Paul Bakke, for their sustained interest in our technology.

MiTi Team

Andrew Z. Hunsberger Said Jahanmir Zhaohui Ren Michael J. Tomaszewski James F. Walton II

Mitsubishi Team

Satoshi Hata Masayuki Kita Other Collaborators Rick Ricker - NIST Petros Sofronis – U of Illinois Brian Sumerday - Sandia

Technical Back-Up Slides

MiTi[®] Compressor Meets DOE Targets

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| Characteristics | Natural Gas Pipelines | DOE Target | MiTi Projection | MiTi Projection |
|---|--------------------------|---------------------------------------|------------------------|----------------------|
| | | | Low Flow | High Flow |
| Efficiency (%) | | 98% | 98% | 98% |
| Hydrogen Capacity Range (kg/day) | | 100,00 | 00 to 1,000,000 | |
| Hydrogen Capacity Target (kg/day) | | 200,000 240,000 500,04 | | |
| Hydrogen Leakage | | <0.5 | 0.2 | 0.2 |
| Hydrogen Purity | | 99.99 | 99.99 | 99.99 |
| Inlet Pressure (psig) | 300-700 | 300-700 | 350 | 500 |
| Discharge Pressure (psig) | 1,000-1,200 | 1,000-1,200 | 1,285 | 1,226 |
| Compressor Component Cost (\$Million) | \$9.2 ¹ | \$9.0 | \$4.1 | \$6.1 |
| Compressor Drive \$400/HP (\$Million) | \$6.4 | | \$3.2 | \$6.4 |
| Total Compressor Package (\$Million) | \$15.6 | | \$7.3 | \$12.5 |
| Maintenance Cost (% total Capital | | | | |
| Investment) | 9.3% ² | 3% | <3% | <3% |
| Maintenance Cost (\$/kW-hr) | \$0.0157 ³ | \$0.007 | <\$0.005 | <\$0.005 |
| Package Size (sq-ft) | ~1,000 | 300-350 | 145 | 160 |
| Reliability (# Systems Required) | | High - Eliminate Redundant Systems | Very High - Modular | - Oil-Free System |
| 1 Oil & Gas Journal, Vol. 107.Issue_34,2010, Transportation. Special Report: Pipeline Profits Capacity Expansion Plans Grow Despite Increased Costs | | | | |

2 DOT/PRCI Pipeline R&D Forum December 11-12, 2003, Washington, DC;

3 Oil & Gas Pipeline Sept 14, 2009, pp77-79

Strength and Fatigue Requirements



Single Stage Test – ASME PTC-10

Type 1 Test: A test conducted with the specified gas at or very near the specified operating conditions.

Type 2 Test: A Test Conducted Subject To The Permissible Deviations Listed In Table Below with Similitude Gas

| Quantity | Symbol | Design Performance | PTC-10 Test Parameters |
|-----------------------|--------------------------------|-----------------------|---------------------------|
| Specific Volume Ratio | v _i /v _d | 1.072 | 1.018 – 1.126 |
| Flow Coefficient | Φ | 0.1253 | 0.120 - 0.130 |
| Machine Mach No. | Mn | 0.3266 | 0.141 - 0.532 |
| Machine Reynolds No. | Re _m | 1.55e6 | 1.55e5 – 1.55e7 |

Test Gas Choices

Hydrogen

Safety Issues & Facility Requirements Beyond Present Scope

Air

- Readily Available, But Drastic Density Variance
- Preliminary Checkout Below Full Speed
 - System Operation
 - Instrumentation
 - Stress/Loading
- Helium
 - Affordable Similitude Gas
 - Full Speed Aerodynamic Validation
 - Qualifies for PTC-10 Type 2 Test

Type 2 Single Stage Test: Helium

| Quantity | Symbol | Test Rig Performance | PTC-10 Test Parameters |
|-----------------------|--------------------------------|-------------------------|---------------------------|
| Specific Volume Ratio | v _i /v _d | 1.052 | 1.018 – 1.126 |
| Flow Coefficient | Φ | 0.1253 | 0.120 - 0.130 |
| Machine Mach No. | Mn | 0.3266 | 0.141 - 0.532 |
| Machine Reynolds No. | Re _m | 3.33e5 | 1.55e5 – 1.55e7 |

Type 2 Test Conditions

- Speed 39,323 rpm (70% of design)
- Inlet Pressure 100 psig
- Inlet Temperature 100°F
- Input Power 137 Hp

- Full Speed Testing Conditions
 - Speed 56,414 rpm
 - Inlet Pressure 50 psig
 - Inlet Temperature 100°F
 - Input Power 260 Hp