Innovative Development, Selection and Testing to Reduce Cost and Weight of Materials for BOP Components

Chris San Marchi
Jonathan Zimmerman
Sandia National Laboratories

DOE Hydrogen and Fuel Cells Program Annual Merit Review
June 9, 2015

Project ID# ST113

This presentation does not contain any proprietary, confidential, or otherwise restricted information
Overview

Timeline
- Project start date: July 2014
- Project end date: Sept 2017

Technical Barriers
A. System Weight and Volume
B. System Cost
H. Balance-of-Plant (BOP) Components

Budget
- Total Project Budget: $2.475M (3yr)
  - Total Federal Share: $2.4M
  - Total Partner Share: $75K
  - Total DOE Funds Spent: $0.3M

Partners
- **Hy-Performance Materials Testing**
  - Subcontractor: fatigue evaluation in hydrogen
- **Swagelok Company**
  - In-kind: materials, test specimens, design perspective
- **Carpenter Technology**
  - In-kind: materials manufacturing expertise
Relevance and Motivation

Problem: BOP components onboard light-duty vehicles collectively dominate cost of the hydrogen storage system at low volumes

- BOP items are a significant fraction of the fuel system costs, even as production volumes increase
- Metallic components (valves, bosses, manifolds, etc) are typically manufactured from “expensive” materials
  - Type 316L (premium grade stainless steel)
  - Low-strength condition (requires thicker walls, driving cost and weight)

Source: DOE Fuel Cell Technologies Office Record # 13010
## Objective

Identify alternative to high-cost metals for high-pressure BOP components

### Barrier from 2012 Storage MYRDD

<table>
<thead>
<tr>
<th>A. System Weight and Volume</th>
<th>Project Goal</th>
</tr>
</thead>
<tbody>
<tr>
<td>Reduce weight by 50%</td>
<td></td>
</tr>
<tr>
<td>Weight can be reduced by optimization of structural stresses</td>
<td></td>
</tr>
</tbody>
</table>

<table>
<thead>
<tr>
<th>B. System Cost</th>
<th>Project Goal</th>
</tr>
</thead>
<tbody>
<tr>
<td>Reduce cost by 35%</td>
<td></td>
</tr>
<tr>
<td>Cost can be reduced by selecting lower cost materials and using less material</td>
<td></td>
</tr>
</tbody>
</table>

<table>
<thead>
<tr>
<th>H. Balance-of-Plant (BOP) Components</th>
<th>Project Goal</th>
</tr>
</thead>
<tbody>
<tr>
<td>Expand the scope of materials of construction for BOP</td>
<td></td>
</tr>
<tr>
<td>Appropriate materials should be determined by relevant performance metrics such as fatigue properties</td>
<td></td>
</tr>
</tbody>
</table>
**Project Approach**

**Objective:** Identify low-cost, light-weight alternatives to annealed type 316L austenitic stainless steels

- *Reduced nickel* content is prime candidate for *cost reduction*
- *High-strength* is prime candidate for *weight reduction*

Two parallel paths:

1. *Experimentally* evaluate fatigue properties of commercial austenitic stainless steels in hydrogen environments
   - Benchmark existing “standard”: annealed type 316L
   - Evaluate alloys with lower-nickel content in high-strength condition

2. *Computational* materials discovery
   - Correlate stacking fault energy (SFE) with hydrogen effects
   - Develop high-throughput computational strategy to determine SFE
   - Use computational strategy to explore alloy additions to increase SFE

**Integration:** Fabricate and measure fatigue performance (experimental) of new alloy combinations (computationally defined)
Project Approach

Simple analysis suggests significant cost and weight reductions can be realized

- Relative component cost is estimated from the relative weight of material and material cost
  - Relative weight is determined from required thickness of material
  - Relative material cost is conservatively informed from price of bar material

\[
t = \frac{PD}{2(SE + PY)} \quad \text{ASME design equation}
\]

<table>
<thead>
<tr>
<th>material</th>
<th>Relative material cost</th>
<th>Yield strength (MPa)</th>
<th>Relative weight</th>
<th>Relative material cost for component</th>
</tr>
</thead>
<tbody>
<tr>
<td>316L</td>
<td>1.0</td>
<td>140</td>
<td>1.0</td>
<td>1.0</td>
</tr>
<tr>
<td>304L</td>
<td>0.84</td>
<td>140</td>
<td>1.0</td>
<td>0.84</td>
</tr>
<tr>
<td>CW 304L</td>
<td>1.7</td>
<td>345</td>
<td>0.46</td>
<td>0.78</td>
</tr>
<tr>
<td>XM-11</td>
<td>0.79</td>
<td>345</td>
<td>0.46</td>
<td>0.36</td>
</tr>
<tr>
<td>CW XM-11</td>
<td>1.6</td>
<td>620</td>
<td>0.17</td>
<td>0.27</td>
</tr>
<tr>
<td>CW XM-19</td>
<td>2.5</td>
<td>725</td>
<td>0.15</td>
<td>0.38</td>
</tr>
</tbody>
</table>
Project Approach

Most hydrogen compatibility decisions are made based on tensile data

- Acceptance metrics from tensile data are undefined/over-specified
  - Strength is unchanged by hydrogen
  - Ductility is decreased by hydrogen

**Ductility is not used as a design parameter**
Project Approach (experimental)

Use stress-based fatigue method for hydrogen from the public domain (CSA CHMC1)

For annealed stainless steels:
- Hydrogen effects do not appear to change the general relationship between yield and the limiting fatigue stress

For high-strength stainless:
- Fatigue may limit practical design stresses

Stress-based fatigue life is used to design pressure systems
- Relevant performance metric and design parameter
Project Approach (computational)

Density functional theory (DFT) enables prediction of fundamental characteristics that correlate with hydrogen effects.

- Implement software needed to interface VASP and Dakota to estimate SFE
- Quantify uncertainties in these calculations
- Intelligently explore composition ‘space’

Use SFE database to develop computationally inexpensive surrogate models and a model design tool.
## Project Approach and Milestones

<table>
<thead>
<tr>
<th>Milestone</th>
<th>Target date</th>
<th>Status</th>
</tr>
</thead>
<tbody>
<tr>
<td>Fatigue life measurements at low temperature (baseline material)</td>
<td>FY15Q2</td>
<td>High-strength alloy selected for initial testing (70% complete)</td>
</tr>
<tr>
<td>Fatigue life measurements in gaseous hydrogen (baseline material)</td>
<td>FY15Q3</td>
<td>Testing started at HPMT (25% complete)</td>
</tr>
<tr>
<td>VASP calculations for Ni and for Fe-Cr-Ni</td>
<td>FY15Q2</td>
<td>Predictions for Ni are consistent with literature (50% complete)</td>
</tr>
<tr>
<td>Comprehensive review of the literature to quantify relationship between measured hydrogen-affected mechanical properties and SFE using regression and correlation analysis</td>
<td>FY15Q4</td>
<td>Data from literature is incomplete</td>
</tr>
</tbody>
</table>

**Go/No Go**

Demonstrate potential for 35% reduction of cost and 50% reduction of weight through the use of alternative commercial alloys or computational alloy design

| FY16                                                                 | XM-11 commercial alloy selected for experimental evaluation; initial testing started (5% complete) |
Accomplishment (experimental)
Baseline fatigue performance established for high-strength type 316L

- High fatigue stress can be achieved with cycles to failure greater than 10,000 cycles (200 years of weekly filling)
- Broader evaluation of performance requires testing at low temperature

Baseline fatigue performance established for high-strength type 316L

- Strain-hardened type 316L
- Ni = 12.04 wt%
- $\sigma_y = 589$ MPa
  - $\sigma_A = 265$ MPa ($R = 0.1$)
- $\sigma_u = 967$ MPa

Notched tension-tension fatigue

- $K_t = 3$, $f = 1$ Hz, $R = 0.1$

- 293K
- 293K, H-precharged

10 Hz

Stress Amplitude (MPa)

Cycles to failure

1000 10^4 10^5 10^6 10^7
Low-temperature fatigue life is “as good as or better” than fatigue life at room temperature.

Broader evaluation of methodology requires testing in gaseous hydrogen at low temperature.

Accomplishment (experimental)

Low-temperature results show non-limiting performance.

Strain-hardened type 316L

- Ni = 12.04 wt%
- Sy = 589 MPa
- $S_A = 265$ MPa ($R = 0.1$)
- Su = 967 MPa

Notched tension-tension fatigue

$K_I = 3$, $f = 1$ Hz, $R = 0.1$

- 223 K
- 223 K, H-precharged
Accomplishment (experimental)

Fatigue life testing in gaseous hydrogen has begun

- Hy-Performance Materials Testing (HPMT) is performing fatigue tests in gaseous hydrogen at pressure of 10 MPa
- HPMT has demonstrated low-temperature tests in gaseous hydrogen for other configurations

Notched tension-tension fatigue

- Ni = 12.04 wt%
- Sy = 589 MPa
  - $S_A = 265$ MPa ($R = 0.1$)
- Su = 967 MPa
Accomplishment (computational)

Ab Initio Calculation of Stacking Fault Energy

- Quantified SFE for fcc Ni using supercell geometries
  - Value is consistent with known literature
  - Value is not sensitive to local magnetic moment
- Assessed computational effort for ternary (Fe-Cr-Ni) stainless steel alloy
  - 450 atoms per supercell needed to ensure system symmetries and small variations in total energies
  - SFE values are sensitive to magnetic moment, resulting in long energy relaxation times
Collaborations and Partnerships

- **Sandia National Laboratories**
  - Core DOE capability for high-pressure hydrogen testing
  - Leverage between NNSA and EERE customers
  - Deep expertise in mechanical metallurgy of austenitic stainless steels
  - Advanced computing tools

- **Hy-Performance Materials Testing (Kevin Nibur)**
  - Commercial testing expertise in pressure environments
  - Unique capabilities in the US

- **Swagelok Company (Shelly Tang)**
  - Component manufacturer
  - Materials selection and engineering analysis
  - Deep understanding of manufacturing with austenitic stainless steels

- **Carpenter Technology (Sam Kernion)**
  - Steel manufacturer
  - Metallurgical expertise and cost analysis
Remaining Challenges and Barriers

- **Challenge**: Fatigue testing at low frequency requires long time (3 days ~ 250K cycles at 1 Hz).
  - **Resolution**: Focus on high stresses, i.e., cycles to failure of 10,000-30,000 cycles

- **Challenge**: Unclear whether existing literature will provide clarity on correlations between SFE, mechanical properties and HE-resistance.
  - **Resolution**: Focus effort on establishing correspondence between relative value and ordering of SFE for various alloy compositions, and known mechanical behavior from experimental side of project and engineering literature.

- **Challenge**: Currently examining extent to which temperature-related contributions to free energy affect SFE values. If influence is significant, high throughput nature of calculations may be compromised.
  - **Resolution**: Use simple compositions to establish the magnitude of this effect, and its computational cost/speed relative to the overall calculations.
Proposed Future Work

Remainder of FY15:

- Complete testing of 316L (benchmark) and commence testing of XM-11 (low-nickel alloy)
- **Go/No Go**: Demonstrate fatigue life test method (CSA CHMC1) for high-pressure hydrogen environments
- Perform transmission electron microscopy (TEM) and analysis to quantify SFE values for select stainless steel alloys: *experimental validation of computations*
  - 316L
  - Fe-Cr-Ni-Mn-Al austenitic stainless steel alloys: IJHE 38 (2013) 9935-9941
  - XM-11 (Fe-21Cr-6Ni-9Mn austenitic stainless steel)
    - excellent candidate but known to be susceptible to hydrogen in tensile tests

*TEM images showing dislocation microstructure in Fe-13Cr-8Ni-10Mn-2.5Al alloy*

*alloy provided by Naumann (BMW) and Michler (Adam Opel/GM)*
Proposed Future Work

Remainder of FY15:

- Comprehensive review of the literature to determine if a correlation exists between SFE and experimentally measured effects of hydrogen on mechanical properties
- Computationally quantify SFE for commercial alloys and Fe-Cr-Ni-Mn-Al alloys
  - 316L, XM-11, Fe-13Cr-8Ni-10Mn-2.5Al
  - Include temperature effects through magnetic entropy contribution to energies
- Develop space-filling sampling strategy to explore effects of different configurations with the same composition on stacking fault energy (SFE)
- Explore permutation techniques to make baseline samples consistent with target composition

*Use Monte Carlo approach to generate a sample of configurations that ensures confidence that the sample size is sufficient.*

- **Go/No Go**: Quantitatively predict the SFE for 3 tertiary compositions relevant to commercial austenitic stainless steels
Proposed Future Work

FY16:

- Establish quantitative comparison of experimental fatigue performance between benchmark and low-nickel alloys
- Create software infrastructure to optimize alloy composition and robustness tradeoffs. Perform prototype studies to compare candidate approaches
- Perform analysis of calculated compositions to quantify trends in estimated SFE and uncertainty. Use Carpenter feedback to extend database on SFE and composition
- **Go/No Go:** Identify one or more candidate materials that potentially meet 35% reduction of cost and 50% reduction of weight using alternative commercial alloys or computational alloy design
Summary

- “Back-of-the-envelope” calculations show large opportunity space for reducing cost and weight of materials for BOP
- Fatigue performance has been benchmarked with:
  - Notched tension-tension fatigue tests (CSA CHMC1)
  - High-strength type 316L with 12 wt% nickel
- Low-temperature fatigue performance suggests limiting behavior may be determined at room temperature for some alloys
- Methodology for \textit{ab initio} determination of SFE is emerging
  - Ni supercell provides values consistent with literature
  - Minimum of 450 atoms per supercell are needed for Fe-Cr-Ni alloys
- TEM and extended fatigue analysis are anticipated to add value to understanding of behaviors and bridging observations at different length scales
Technical Back-Up Slides
Fracture mechanics design using fatigue crack growth is standardized in ASME BPVC VIII.3 KD

Concern: Fatigue crack growth design methodologies have not been implemented for design of manifold components.
Fatigue testing at low frequency requires long testing times

Testing times at 1Hz:
- ~4 hours
- ~3 days
- ~70 days

Graph showing the relationship between stress amplitude (MPa) and cycles to failure, with annotations indicating 'SY of annealed 316'.
Proposed Future Work

FY16: leveraging industrial partners

- Perform preliminary set of optimized calculations and assemble initial version of SFE database. Deliver set to Carpenter Technology Corporation for feedback
- Explore extrapolation of data to
  - design (e.g., collaboration with Swagelok)
  - other fatigue methodologies (e.g., non-notched geometry and crack growth)

As-received
\[ S_A = 200 \text{ MPa} \]

H-precharged
\[ S_A = 190 \text{ MPa} \]

Fatigue fracture surfaces
Test temperature = -50 °C

As-received
\[ 0.7 \mu m/\text{cycle} \]

H-precharged
\[ 0.6 \mu m/\text{cycle} \]